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Dynamics of delayed piecewise linear systems *

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Abstract

In this paper the dynamics of the controlled pendulum is investigated assuming backlash and time delays. The upper equilibrium of the pendulum is stabilized by a piecewise constant control force which is the linear combination of the sampled values of the angle and the angular velocity of the pendulum. The control force is provided by a motor which drives one of the wheels of the cart through an elastic teeth belt. The contact between the teeth of the gear (rigid) and the belt (elastic) introduces a nonlinearity known as "backlash" and causes the oscillation of the controlled pendulum around its upper equilibrium. The processing and sampling delays in the determination of the control force tend to destabilize the controlled system as well. We obtain conditions guaranteeing that the pendulum remains in the neighborhood of the upper equilibrium. Experimental findings obtained on a computer controlled inverted pendulum cart structure are also presented showing good agreement with the simulation results.

1 Introduction

Dynamical systems with piecewise linear components in the equations of motion occur frequently in practice. Gear pairs with backlash [23], railway wheelsets with clearance [13], mechanical oscillators with clearance [6, 14, 15, 19] or amplitude constraints [20, 21], impact dampers [16, 17, 18], moving parts with dry friction [12] and adjacent structures during earthquake [4, 5] are modelled by systems with piecewise linear stiffness, damping or restoring force.

Control is often added to such systems to stabilize unstable equilibria. Digital balancing of the inverted pendulum is examined in the following sections. The pendulum is placed on a cart and controlled by a motor which drives one of the wheel-axle of the cart by an elastic teeth belt [2, 7]. Backlash occurring at the driving wheel of the motor makes the system piecewise linear and creates an unstable zone around the otherwise stable equilibrium of the controlled system. Consequently, the best possible outcome of the stabilization process is to keep the pendulum in a small neighborhood of its upper equilibrium.

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Figure 2.1: The inverted pendulum on a cart and its stability map

In Section 2, a model of the inverted pendulum on a cart is derived. Stability analysis is performed using the assumption that the belt is perfectly elastic, (i.e. no backlash). Stability chart for control parameters is also obtained and parameterized by the stiffness of the driving belt. In Section 3, backlash at the contact of the driving belt and the axle of the motor is considered. The effect of backlash and time delay together is investigated in Section 4. Experimental observations are presented in Section 5.

2 The inverted pendulum on a cart

The model of the inverted pendulum on a cart can be seen in Figure 2.1, [8, 9, 10, 11]. The cart can move only in the horizontal direction. The motor drives one of the wheels of this cart through a driving-belt with stiffness s. The system has three degrees of freedom described by the general coordinates q_1, q_2, q_3 , denoting the displacement x of the cart, the angle φ of the pendulum and the angle ψ of the motor axle, respectively. It is assumed that the angle φ of the pendulum and the displacement x of the cart are observed together with their derivatives and the observed values provide the motor voltage U_m according to the equation

$$U_m = P_1 x + D_1 \dot{x} + P_2 \varphi + D_2 \dot{\varphi}, \qquad (2.1)$$

where the constants P_1, D_1, P_2, D_2 are the so-called PD controller parameters. U_m together with $\dot{\psi}$, the angular velocity of the motor determine M_m , the driving torque of the motor by

$$M_m = LU_m - K\psi, \qquad (2.2)$$

where L and K are given motor parameters.

The nonlinear equations of motion are obtained (see also Figure 2.1) by means of the Lagrange equations of the second kind. The kinetic energy \mathcal{T} , the

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potential \mathcal{U} , the dissipation \mathcal{D} and the general force Q have the following form

$$\mathcal{T} = \frac{1}{2}m\left(\left(\dot{x} + \frac{l}{2}\dot{\varphi}\cos\varphi\right)^{2} + \left(\frac{l}{2}\dot{\varphi}\sin\varphi\right)^{2}\right) + \frac{1}{24}ml^{2}\dot{\varphi}^{2} + \frac{1}{2}M\dot{x}^{2} + \frac{1}{2}J_{m}\dot{\psi}^{2},$$
$$\mathcal{U} = -\frac{1}{2}mgl\left(1 - \cos\varphi\right) + U_{s},$$
$$\mathcal{D} = 0, \quad Q_{1}^{e} = 0, \quad Q_{2}^{e} = 0, \quad Q_{3}^{e} = M_{m},$$
(2.3)

where the index *e* refers to elastic, $U_s = 0$ if the belt is ideally rigid, $U_s = s\Delta^2/2$ if the belt is elastic and Δ is the elongation of the spring, $M = m_m + m_c + 4m_w + 4J_w/R_w^2$, the sum of the mass of the motor m_m , the cart m_c and the reduced mass of inertia of the wheels $4m_w + 4J_w/R_w^2$. Applying the Lagrange equations of the second kind

$$\frac{\mathrm{d}}{\mathrm{d}t}\frac{\partial \mathcal{T}}{\partial \dot{q}_k} - \frac{\partial \mathcal{T}}{\partial q_k} + \frac{\partial \mathcal{U}}{\partial q_k} + \frac{\partial \mathcal{D}}{\partial \dot{q}_k} = Q_k \,, \qquad (2.4)$$

k = 1, 2, 3, the nonlinear equations of motion for the controlled pendulum are obtained

$$\begin{pmatrix} m+M & \frac{1}{2}ml\cos\varphi & 0\\ \frac{1}{2}ml\cos\varphi & \frac{1}{3}ml^2 & 0\\ 0 & 0 & \frac{1}{2}m_mr_m^2 \end{pmatrix} \begin{pmatrix} \ddot{x}\\ \ddot{\varphi}\\ \ddot{\psi} \end{pmatrix} + \begin{pmatrix} s\frac{r_w^2}{R_w^2} & 0 & -sr_m\frac{r_w}{R_w}\\ 0 & 0 & 0\\ -sr_m\frac{r_w}{R_w} & 0 & sr_m^2 \end{pmatrix} \begin{pmatrix} x\\ \varphi\\ \psi \end{pmatrix} + \begin{pmatrix} -\frac{1}{2}ml\dot{\varphi}^2\sin\varphi\\ -\frac{1}{2}mgl\sin\varphi\\ 0 \end{pmatrix} = \begin{pmatrix} 0\\ 0\\ Q_3^e \end{pmatrix}.$$
(2.5)

The case when the belt is ideally rigid

If the belt is ideally rigid [8], then x and ψ are related by the equation $\psi = xr_w/(r_mR_w)$, where r_w and R_w are radii of the wheel and r_m is the radius of the motor axle and then ψ can be eliminated from equation (2.5). Multiplying the third equation of (2.5) by $r_m r_w/R_w$, adding it to the first equation and substituting $\psi = xr_w/(r_mR_w)$ yields the nonlinear equations of motion

$$\begin{pmatrix} m_r & \frac{1}{2}ml\cos\varphi\\ \frac{1}{2}ml\cos\varphi & \frac{1}{3}ml^2 \end{pmatrix} \begin{pmatrix} \ddot{x}\\ \ddot{\varphi} \end{pmatrix} - \begin{pmatrix} \frac{1}{2}ml\dot{\varphi}^2\sin\varphi\\ \frac{1}{2}mgl\sin\varphi \end{pmatrix} = \begin{pmatrix} Q_1^r\\ 0 \end{pmatrix}, \quad (2.6)$$

where $m_r = m + M + m_m r_w^2/(2R_w^2)$, m and l are the mass and the length of the pendulum and $Q_1^r = Q_3^e r_w/(r_m R_w)$ with index r referring to rigid. Linearizing these equations around the upper equilibrium $\varphi = 0$ we obtain the variational system

$$\begin{pmatrix} m_r & \frac{1}{2}ml\\ \frac{1}{2}ml & \frac{1}{3}ml^2 \end{pmatrix} \begin{pmatrix} \ddot{x}\\ \ddot{\varphi} \end{pmatrix} - \begin{pmatrix} 0 & 0\\ 0 & \frac{1}{2}mgl \end{pmatrix} \begin{pmatrix} x\\ \varphi \end{pmatrix} = \begin{pmatrix} Q_1^r\\ 0 \end{pmatrix}.$$
 (2.7)

Using equations (2.1), (2.2) and (2.3), the control force has the form

$$Q_1^r = \frac{r_w}{r_m R_w} L \left(P_1 x + D_1 \dot{x} + P_2 \varphi + D_2 \dot{\varphi} \right) - K \frac{r_w^2}{r_m^2 R_w^2} \dot{x} \,. \tag{2.8}$$

It is easy to see from (2.8) that if

$$P_1 = 0 \text{ and } D_1 = \frac{1}{L} \frac{r_w}{r_m R_w} K,$$
 (2.9)

then Q_1^r can be written as

$$Q_1^r = \frac{r_w}{r_m R_w} \left(P\varphi + D\dot{\varphi} \right) \,, \tag{2.10}$$

where $P = LP_2$ and $D = LD_2$. Thus, the control force Q_1^r is only a function of φ and $\dot{\varphi}$ (i.e., x and \dot{x} do not appear in Q_1^r). The displacement x of the cart can then be eliminated from the equations of motion as follows. Solve the first equation of (2.7) for \ddot{x} and substitute the given expression into the second equation of (2.7). A one degree of freedom system is obtained with the following single second order governing equation

$$\ddot{\varphi} - \frac{6g}{l}\varphi = -\frac{6}{m_r l}Q_1^r.$$
(2.11)

In our setting stabilization means that φ and $\dot{\varphi}$ are both zero, but x and \dot{x} could be nonzero. Due to the relationship between \ddot{x} , φ , $\dot{\varphi}$ and $\ddot{\varphi}$ as described by equation (2.6), we get that \dot{x} is constant when $\varphi = \dot{\varphi} = 0$. (I.e., the cart is moving with constant velocity or possibly standing.)

We have the following result concerning finding suitable control parameters P, D in (2.10).

Theorem 2.1 The trivial solution of (2.10)-(2.11) is asymptotically stable if and only if

$$P > P_0 = m_r g r_m \frac{R_w}{r_w} \quad and \quad D > 0.$$

$$(2.12)$$

Proof: The characteristic equation of (2.11) has the following form

$$\lambda^2 + \frac{6r_w}{m_r lr_m R_w} D\lambda + \left(\frac{6r_w}{m_r lr_m R_w} P - \frac{6g}{l}\right) = 0.$$
(2.13)

The coefficients in (2.13) are clearly positive if the conditions (2.12) are satisfied. The statement of the theorem follows.

The case when the belt is elastic

If the belt is elastic, then the nonlinear equations of motion assume the form of (2.5). This system can be reduced to a two degrees of freedom system if the assumptions (2.9) are applied again and a new general coordinate, Δ , the elongation of the spring is introduced by

$$\Delta = r_m \psi - \frac{r_w}{R_w} x \,. \tag{2.14}$$

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Multiplying the first and third equation of (2.5) by $(-m_m r_m r_w)/(2R_w)$ and (m+M), respectively, summing them and using (2.14) gives the first equation of motion of the reduced system. Solving the first equation of (2.5) for \ddot{x} , using (2.14) and substituting it into the second equation of (2.5) gives the second equation of motion of the reduced system and altogether we have

$$\begin{pmatrix} \frac{1}{2} \left(m+M\right) m_m r_m & -\frac{1}{4} m m_m l r_m \frac{r_w}{R_w} \cos \varphi \\ 0 & \frac{1}{3} m l^2 - \frac{1}{4} \frac{m}{(m+M)} m l^2 \cos^2 \varphi \end{pmatrix} \begin{pmatrix} \ddot{\Delta} \\ \ddot{\varphi} \end{pmatrix}$$

$$+ \begin{pmatrix} \left(m+M\right) \frac{K}{r_m} & 0 \\ 0 & 0 \end{pmatrix} \begin{pmatrix} \dot{\Delta} \\ \dot{\varphi} \end{pmatrix} + \begin{pmatrix} \frac{1}{4} m m_m l r_m \frac{r_w}{R_w} \dot{\varphi}^2 \sin \varphi \\ \frac{1}{4} \frac{m}{(m+M)} m l^2 \dot{\varphi}^2 \sin \varphi \cos \varphi - \frac{1}{2} m g l \sin \varphi \end{pmatrix}$$

$$+ \begin{pmatrix} \left(m+M\right) r_m + \frac{1}{2} m_m r_m \frac{r_w^2}{R_w^2} \\ \frac{1}{2} \frac{m}{(m+M)} l \frac{r_w}{R_w} \end{pmatrix} R_s = \begin{pmatrix} \left(m+M\right) Q \\ 0 \end{pmatrix} ,$$

$$(2.15)$$

where

$$R_s = \begin{pmatrix} -s \frac{r_w}{R_w} & 0 & sr_m \end{pmatrix} \begin{pmatrix} x\\ \varphi\\ \psi \end{pmatrix} = s\Delta$$
(2.16)

is the force in the driving belt and

$$Q = P\varphi + D\dot{\varphi} \tag{2.17}$$

is the control force. Linearizing these equations around the upper equilibrium $\varphi=0$ yields

$$\begin{pmatrix} \frac{1}{2} (m+M) m_m r_m & -\frac{1}{4} m m_m l r_m \frac{r_w}{R_w} \\ 0 & \frac{1}{3} m l^2 - \frac{1}{4} \frac{m}{(m+M)} m l^2 \end{pmatrix} \begin{pmatrix} \ddot{\Delta} \\ \ddot{\varphi} \end{pmatrix} \\ + \begin{pmatrix} (m+M) \frac{K}{r_m} & -(m+M) D \\ 0 & 0 \end{pmatrix} \begin{pmatrix} \dot{\Delta} \\ \dot{\varphi} \end{pmatrix}$$
(2.18)
$$+ \begin{pmatrix} (m+M) r_m s + \frac{1}{2} m_m r_m \frac{r_w^2}{R_w^2} s & -(m+M) P \\ \frac{1}{2} \frac{m}{(m+M)} l \frac{r_w}{R_w} s & -\frac{1}{2} m g l \end{pmatrix} \begin{pmatrix} \Delta \\ \varphi \end{pmatrix} = \begin{pmatrix} 0 \\ 0 \end{pmatrix}.$$

The subsequent theorem establishes the stability properties of system (2.18).

Theorem 2.2 The trivial solution of system (2.18) is asymptotically stable if and only if

$$P > P_0 = m_r g r_m \frac{R_w}{r_w}, \quad s > s_{cr} = \frac{3(m+M)m_m g}{(m+4M+2m_m r_w^2/R_w^2)l}, \quad q(P,D,s) < 0,$$
(2.19)

where $q(P, D, s) = a_{22}D^2 + a_1P + a_2D + a_0$ is a parabola, with the coefficients

$$\begin{split} &a_{22} = 6m_m r_w R_w r_m^3 s \,, \\ &a_1 = 2 \,(m+4M) \, l R_w^2 K^2 \,, \\ &a_2 = -6 \,(m+M) \, m_m g R_w^2 r_m^2 K - 2 \,(m+4M) \, l R_w^2 r_m^2 K s - 4m_m l r_w^2 r_m^2 K s \,, \\ &a_0 = 3m m_m g l r_w R_w r_m K^2 \,. \end{split}$$

Proof: The characteristic equation of (2.18) can be written as

$$b_4\lambda^4 + b_3\lambda^3 + b_2\lambda^2 + b_1\lambda + b_0 = 0, \qquad (2.20)$$

where

$$\begin{split} b_4 &= \frac{1}{24} \left(m + 4M \right) m_m l r_m^2, \\ b_3 &= \frac{1}{12} \left(m + 4M \right) l K, \\ b_2 &= \frac{1}{12} \left(m + 4M + 2m_m \frac{r_w^2}{R_w^2} \right) l r_m^2 s - \frac{1}{4} \left(m + M \right) m_m g r_m^2, \\ b_1 &= \frac{1}{2} r_m \frac{r_w}{R_w} s D - \frac{1}{2} \left(m + M \right) g K, \\ b_0 &= \frac{1}{2} r_m \frac{r_w}{R_w} s P - \frac{1}{2} \left(m + M + \frac{1}{2} m_m \frac{r_w^2}{R_w^2} \right) g r_m^2 s \,. \end{split}$$

The Hurwitz matrix assumes the form

$$\begin{pmatrix} b_1 & b_0 & 0 & 0 \\ b_3 & b_2 & b_1 & b_0 \\ 0 & b_4 & b_3 & b_2 \\ 0 & 0 & 0 & b_4 \end{pmatrix}.$$

According to the Routh-Hurwitz criterion [3], the statement of the theorem holds if all the coefficients of the characteristic equation and all the leading principal minors of the Hurwitz matrix, i.e. the Hurwitz determinants, are positive. The conditions $b_3 > 0$ and $b_4 > 0$ hold because b_3 and b_4 depend only on positive physical parameters. The conditions $b_0 > 0, b_2 > 0$ and $H_3 = b_1b_2b_3 - b_1^2b_4 - b_0b_3^2 > 0$ are equivalent to $P > P_0, s > s_{cr}$ and q(P, D, s) < 0, respectively. The positivity of the Hurwitz determinants $H_1 = b_1$ and $H_2 = b_1b_2 - b_0b_3$ follow from the positivity of the coefficients b_0, b_2, b_3, b_4 and the Hurwitz determinant H_3 , because these conditions imply that $b_1b_2 - b_0b_3 > b_1^2b_4/b_3 > 0$ and $b_1 > b_0b_3/b_2 > 0$, i.e., $H_2 > 0$ and $H_1 > 0$. \Box

In the remainder of this section first we study the dynamic behavior of the linearized model of the controlled pendulum on the boundary of the stability region determined in Theorem 2.2, and then we present a numerical bifurcation analysis for the original nonlinear system (2.15).

The stability charts in the plane of the control parameters are shown in Figure 2.1 for rigid and elastic belts. The stability domain for rigid belt is a half plane and for the elastic belt it is bordered by a line and a parabola. A real characteristic root changes its sign crossing the line, while a complex conjugate pair turns up in the right half of the complex plane crossing the parabola q(P, D, s) = 0.

The angular frequency, ω , of the oscillation at the loss of stability is the imaginary part of the characteristic roots with zero real parts, thus it can be derived by substituting $\lambda = i\omega$ in the characteristic equation (2.20) and solving

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Figure 2.2: (a) The frequency of the oscillation and the stability domain (b) Bifurcation diagrams for D = 2 [Nms] (upper) and for P = 20 [Nm] (lower)

either its real or imaginary part for ω . Using the fact that q(P, D, s) = 0 we can obtain the frequency of the oscillation as a function of P or D only. For example, ω as a function of P can be obtained as the positive solution of the equation

$$\frac{1}{24} (m+4M) m_m lr_m^2 \omega^4 - \left(\frac{1}{12} \left(m+4M+2m_m \frac{r_w^2}{R_w^2}\right) lr_m^2 s - \frac{1}{4} (m+M) m_m gr_m^2\right) \omega^2 + \frac{1}{2} r_m \frac{r_w}{R_w} sP - \frac{1}{2} \left(m+M+\frac{1}{2}m_m \frac{r_w^2}{R_w^2}\right) gr_m^2 s = 0$$

The frequency $f = \omega/(2\pi)$ of the oscillation at the loss of stability and the stability domain are drawn in Figure 2.2(a) for the given values of parameters: m = 0.169 [kg], M = 1.136 [kg], $m_m = 0.2$ [kg], g = 9.81 [m/s²], l = 0.5 [m], $r_w = 0.02$ [m], $R_w = 0.03$ [m], $r_m = 0.01$ [m], K = 0.01 [Nms] and s = 10000[N/m]. Figure 2.2(a) shows that there is a unique frequency for every pair P, Don the parabola.

To study the relationship between (2.18) and (2.15) we performed a numerical bifurcation analysis using the software package AUTO (available at

ftp://ftp.cs.concordia.ca/pub/doedel/auto/). Results are presented in Figure 2.2(b) for the values of parameters given in the previous paragraph. The bifurcation parameter is P in the upper half of Figure 2.2(b), and the control parameter D is kept at 2 [Nms]. A pitchfork bifurcation is occurred at $P_0 = 0.1986$ [Nm] obtained from (2.19), where the upper equilibrium of the pendulum becomes stable. Figure 2.2(b) indicates that the upper equilibrium maintains its stability till a supercritical Hopf-bifurcation occurs at $P_1 = 139.7$ [Nm] obtained from the condition q(P, 2, 10000) = 0. An unstable stationary solution appears at the pitchfork bifurcation and φ tends to $\pi/2$ along this solution as P increases. A stable periodic solution appears at the supercritical Hopf-bifurcation.

The bifurcation parameter is D in the lower half of Figure 2.2(b), and the control parameter P is kept at 20 [Nm]. The upper equilibrium is stable between the Hopf-bifurcation points. They occur at $D_1 = 0.1991$ [Nms] and $D_2 = 5.916$ [Nms] obtained from the condition q(20, D, 10000) = 0. Periodic solution exists at the border of the stability domain where two complex conjugate characteristic roots are crossing the imaginary axis.

If φ is less than its value at the unstable stationary solution then trajectories tend to the stable equilibrium. It follows that if P is chosen from the stability domain and not very close to P_0 then the upper equilibrium of the pendulum is stabilized even if the initial value of the angle φ is not small. Thus, the linear variational system (2.18) can be considered a good approximation of the original nonlinear system (2.15) in the stability region given by (2.19).

3 Backlash at the gears

Backlash appears in the system as a nonlinearity in the belt-drive. In particular, the force in the belt, R_s , is given by

$$R_s = \begin{cases} s \left(\Delta + r_0\right) & \text{if } \Delta \le -r_0 \\ 0 & \text{if } |\Delta| < r_0 \\ s \left(\Delta - r_0\right) & \text{if } \Delta \ge r_0, \end{cases}$$
(3.1)

where s and Δ is defined as before and r_0 is half of the backlash.

Consequently, the equations of motion (2.15) with (3.1) and control force (2.17), after linearization with respect to φ , yield

$$\begin{pmatrix}
\frac{1}{2}(m+M)m_{m}r_{m} & -\frac{1}{4}mm_{m}lr_{m}\frac{r_{w}}{R_{w}}\\
0 & \frac{1}{3}ml^{2} - \frac{1}{4}\frac{m}{(m+M)}ml^{2}
\end{pmatrix}
\begin{pmatrix}
\ddot{\Delta}\\
\ddot{\varphi}
\end{pmatrix}
+ \begin{pmatrix}
(m+M)\frac{K}{r_{m}} & -(m+M)D\\
0 & 0
\end{pmatrix}
\begin{pmatrix}
\dot{\Delta}\\
\dot{\varphi}
\end{pmatrix}
+ \begin{pmatrix}
(m+M)r_{m}s + \frac{1}{2}m_{m}r_{m}\frac{r_{w}^{2}}{R_{w}^{2}}s & -(m+M)P\\
\frac{1}{2}\frac{m}{m+M}l\frac{r_{w}}{R_{w}}s & -\frac{1}{2}mgl
\end{pmatrix}
\begin{pmatrix}
\Delta\\
\varphi
\end{pmatrix}
= \begin{pmatrix}
((m+M)r_{m}sr_{0} + \frac{1}{2}m_{m}r_{m}\frac{r_{w}^{2}}{R_{w}^{2}}sr_{0}
\end{pmatrix}\operatorname{sgn}\Delta\\
\frac{1}{2}\frac{m}{m+M}l\frac{r_{w}}{R_{w}}sr_{0}\operatorname{sgn}\Delta
\end{pmatrix},$$
(3.2)

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Figure 3.1: (a), (b) Bifurcation diagrams, the bifurcation parameter is P and D, respectively (c) The bifurcation diagram near the point where the computation is interrupted (d) The stability chart with the bifurcation curves

if $|\Delta| \geq r_0$, and

$$\begin{pmatrix} \frac{1}{2} (m+M) m_m r_m & -\frac{1}{4} m m_m l r_m \frac{r_w}{R_w} \\ 0 & \frac{1}{3} m l^2 - \frac{1}{4} \frac{m}{(m+M)} m l^2 \end{pmatrix} \begin{pmatrix} \ddot{\Delta} \\ \ddot{\varphi} \end{pmatrix}$$

$$+ \begin{pmatrix} (m+M) \frac{K}{r_m} & -(m+M) D \\ 0 & 0 \end{pmatrix} \begin{pmatrix} \dot{\Delta} \\ \dot{\varphi} \end{pmatrix}$$

$$+ \begin{pmatrix} 0 & -(m+M) P \\ 0 & -\frac{1}{2} m g l \end{pmatrix} \begin{pmatrix} \Delta \\ \varphi \end{pmatrix} = \begin{pmatrix} 0 \\ 0 \end{pmatrix},$$

$$(3.3)$$

if $|\Delta| < r_0$.

We have calculated exact (Mathematica) and approximate (MATLAB) solutions for (3.2)–(3.3) using the same parameter values as in Section 2, and $r_0 = 0.001$ [m] for the backlash in the system. Our simulations were performed with control parameters chosen from the region given by (2.19). We also performed numerical bifurcation analysis using AUTO. The bifurcation diagrams are shown in Figure 3.1(a) and 3.1(b), where the bifurcation parameters are the control parameters, P and D, respectively. D = 2 [Nms] in Figure 3.1(a) and 3.1(c) (which is an enlargement of a section of Figure 3.1(a)). There is a

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branch point at one of the borders, P_0 , of the stability domain (i.e. the domain given by (2.19)), where stable fix points appear. This is the same value which is determined in Theorem 2.2. A homoclinic orbit occurs at $P = P_{cr}$, so there is a homoclinic bifurcation there, where a limit cycle appears. P_{cr} is approximately 4 according to the AUTO-computations and it is 3.46 according to the MATLAB-simulations. Only the fix points are stable between P_0 and P_{cr} , while all the fix points and the limit cycle are stable for values of P between P_{cr} and P_2 . All of them have a domain of attraction, so trajectories spiral to one of them depending on the initial conditions. There is another bifurcation point at P_2 , where the fix points become unstable. The limit cycle retains its stability with increasing amplitude till the parameter P reaches the other border P_1 of the stability domain.

P = 40 [Nm] in Figure 3.1(b). There is a stable limit cycle between the borders of the stability domain but the fix points are stable only along the continuous line between the bifurcation points indicated with squares in the figure.

According to the results summarized above, the stability domain in the plane of the control parameters is constructed and sketched in Figure 3.1(d). It has the same boundary as in the case of no backlash (i.e. (2.19)). Only the fix points are stable in a small domain near the line. Stable limit cycle appears at the homoclinic bifurcation points indicated with the dotted line. Fix points lose their stability at the other bifurcation points indicated with the smashed line, so all the fix points and the limit cycle are stable between the dotted and the smashed line, and only the limit cycle is stable in the remaining part of the stability domain.

We note that a periodic solution corresponds to the oscillation of the stick around its vertical equilibrium. The physical meaning of the stable equilibria is that the control force does not push the stick further than the vertical line and it oscillates with decreasing amplitude on either side of the vertical position.

4 The inverted pendulum on a cart with time delay

In this section, sampling, $\tau_s > 0$, and processing, $\tau_p > 0$, delays are included in the model of the inverted pendulum on a cart [9], [11]. First, we consider the case of no backlash. Then the linearized equations of motion are given by (2.18) as before, but now delayed arguments will appear due to the fact that the control force, Q(t), takes the form

$$Q(t) = P\varphi((j-1)\tau_s) + D\dot{\varphi}((j-1)\tau_s), \quad t \in [(j-1)\tau_s + \tau_p, j\tau_s + \tau_p),$$

for j = 1, 2, ... We shall assume that the sampling and the processing delays are the same, i.e., $\tau_s = \tau_p = \tau$. The more general case, $\tau_s \neq \tau_p$, could be considered

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similarly. With that, the control force can be written as

$$Q(t) = P\varphi((j-1)\tau) + D\dot{\varphi}((j-1)\tau), \quad t \in [j\tau, (j+1)\tau), \quad j = 1, 2, \dots$$
(4.1)

Without time delay we had a system (2.18) of ordinary differential equations. With control force (4.1) we obtain the following system of delay differential equations

$$\begin{pmatrix} \frac{1}{2} (m+M) m_m r_m & -\frac{1}{4} m m_m l r_m \frac{r_w}{R_w} \\ 0 & \frac{1}{3} m l^2 - \frac{1}{4} \frac{m}{(m+M)} m l^2 \end{pmatrix} \begin{pmatrix} \ddot{\Delta}(t) \\ \ddot{\varphi}(t) \end{pmatrix} \\ + \begin{pmatrix} (m+M) \frac{K}{r_m} & 0 \\ 0 & 0 \end{pmatrix} \begin{pmatrix} \dot{\Delta}(t) \\ \dot{\varphi}(t) \end{pmatrix} \\ + \begin{pmatrix} (m+M) r_m s + \frac{1}{2} m_m r_m \frac{r_w^2}{R_w^2} s & 0 \\ \frac{1}{2} \frac{m}{(m+M)} l \frac{r_w}{R_w} s & -\frac{1}{2} m g l \end{pmatrix} \begin{pmatrix} \Delta(t) \\ \varphi(t) \end{pmatrix} \\ = \begin{pmatrix} (m+M) \left(P \varphi \left((j-1) \tau \right) + D \dot{\varphi} \left((j-1) \tau \right) \right) \\ 0 \end{pmatrix}, \quad t \in [j\tau, (j+1) \tau),$$

for $j = 1, 2, \ldots$. Transform the second order system (4.2) into a system of first order equations, i.e., multiply (4.2) by the inverse of the coefficient matrix of the second order term and introduce a new state vector **y**. We obtain

$$\dot{\mathbf{y}}(t) = \mathbf{A}\mathbf{y}(t) + \mathbf{b}Q(t), \qquad (4.3)$$

where

$$\mathbf{A} = \begin{pmatrix} 0 & 1 & 0 & 0 \\ -\frac{4r_w^2 s}{(m+4M)R_w^2} - \frac{2s}{m_m} & -\frac{2K}{m_m r_m^2} & \frac{3mgr_w}{(m+4M)R_w} & 0 \\ 0 & 0 & 0 & 1 \\ -\frac{6r_w s}{(m+4M)lR_w} & 0 & \frac{6(m+M)g}{(m+4M)l} & 0 \end{pmatrix}, \quad (4.4)$$
$$\mathbf{b} = \begin{pmatrix} 0 \\ \frac{2}{m_m r_m} \\ 0 \\ 0 \end{pmatrix}, \quad \mathbf{y} = \begin{pmatrix} \Delta \\ \dot{\Delta} \\ \varphi \\ \dot{\varphi} \end{pmatrix}.$$

Note that Q(t) can be expressed as

$$Q(t) = \mathbf{Ky}((j-1)\tau), \quad t \in [j\tau, (j+1)\tau), \quad j = 1, 2, \dots$$

where

$$\mathbf{K} = \begin{pmatrix} 0 & 0 & P & D \end{pmatrix}. \tag{4.5}$$

The general solution of (4.3) for $t > t_0$ can be written as

$$\mathbf{y}(t) = e^{\mathbf{A}(t-t_0)}\mathbf{y}(t_0) + \int_{t_0}^t e^{\mathbf{A}(t-s)}\mathbf{b}Q(s) \,\mathrm{d}s.$$
(4.6)

If in (4.6) we select $t_0 = j\tau$, $t = (j+1)\tau$, and introduce the notations $\mathbf{y}_j = \mathbf{y}(j\tau)$, $Q_j = Q(j\tau)$, $j = 1, 2, \dots$, we have

$$\mathbf{y}_{j+1} = e^{\mathbf{A}\tau} \mathbf{y}_j + \int_{j\tau}^{(j+1)\tau} e^{\mathbf{A}((j+1)\tau-s)} \mathbf{b} ds Q_j = e^{\mathbf{A}\tau} \mathbf{y}_j + \int_0^{\tau} e^{\mathbf{A}(\tau-s)} \mathbf{b} ds Q_j, \quad j = 1, 2, \dots$$
(4.7)

Note that in (4.7) we have used that Q(t) is a piecewise constant function. Let

$$\mathbf{A}_{\mathbf{d}} = e^{\mathbf{A}\tau}, \quad \mathbf{b}_{\mathbf{d}} = \int_{0}^{\tau} e^{\mathbf{A}(\tau-s)} \mathrm{d}s\mathbf{b}.$$
(4.8)

The discrete-time model, i.e., $t = j\tau$, j = 1, 2, ..., consisting the state and feedback equations assumes the form

$$\mathbf{y}_{j+1} = \mathbf{A}_{\mathbf{d}} \mathbf{y}_j + \mathbf{b}_{\mathbf{d}} Q_j$$
$$Q_{j+1} = \mathbf{K} \mathbf{y}_j$$

for $j = 1, 2, \ldots$, or equivalently

$$\mathbf{z}_{j+1} = \mathbf{S}\mathbf{z}_j, \quad j = 1, 2, \dots \tag{4.9}$$

where

$$\mathbf{z}_j = \begin{pmatrix} \mathbf{y}_j \\ Q_j \end{pmatrix}, \quad \mathbf{S} = \begin{pmatrix} \mathbf{A}_\mathbf{d} & \mathbf{b}_\mathbf{d} \\ \mathbf{K} & 0 \end{pmatrix}.$$
 (4.10)

Recall that the discrete system (4.9) is asymptotically stable if and only if all its characteristic multipliers μ_i , i = 1, ..., 5 are in modulus less than one, i.e., $|\mu_i| < 1, i = 1, ..., 5$. The characteristic equation of (4.9) has the following form

$$g_5\mu^5 + g_4\mu^4 + g_3\mu^3 + g_2\mu^2 + g_1\mu + g_0 = 0, \qquad (4.11)$$

where the coefficients g_i , i = 0, ..., 5 are determined by the system parameters. Applying the Moebius-Zukovski transformation [3], i.e., substituting $\mu = (\nu + 1) / (\nu - 1)$ into (4.11) and multypling it by $(\nu - 1)^5$, we obtain a fifth order equation for ν in the form of (4.11). Since $|\mu_i| < 1$, i = 1, ..., 5 if and only if $\Re \nu_i < 0$, i = 1, ..., 5, the Routh-Hurwitz criterion can be used to determine the stability conditions following the same procedure as in Theorem 2.2.

The stability domain is bordered by the same line as in case of the system without time delay and a parabola which depends on the time delay, resulting in a shrinking stability domain for increasing time delays. The stability chart is given in Figure 4.1(a) for $\tau = 0$ [ms] and $\tau = 5$ [ms].

The stability domain disappears at a certain critical value of the time delay, so balancing is impossible above that value. The critical delay depends on the parameters describing the system. The most interesting is the dependence on the length of the pendulum and the stiffness of the belt. The connections between the critical time delay and the length of the pendulum and between the critical time delay and the stiffness of the belt are shown in Figure 4.2(a).



Figure 4.1: (a) The stability charts for $\tau = 0$ [ms] and $\tau = 5$ [ms] (b) The domain of 2 kinds of stable motions for $\tau = 5$ [ms]

In the latter case we can see an asymptote which corresponds to the result gained for the system with rigid belt. As the length of the pendulum decreases, the critical time delay also decreases. There is a minimal length l_{min} of the pendulum, where the critical time delay becomes 0. It is proportional to 1/s. If the belt were ideally rigid, then $l_{min} = 0$, but since the belt is elastic, $l_{min} > 0$. Our result is shown in Figure 4.2(a), where s = 10 [kN/m] and $l_{min} = 0.16$ [mm]. Similarly, as the stiffness of the belt decreases, the critical time delay also decreases and at a certain minimal value s_{min} , it becomes 0. The minimal stiffness of the belt is proportional to 1/l. In case of Figure 4.2(a), l = 0.5 [m] and $s_{min} = 3.14$ [N/m].

Now we study the behavior of system (4.9) on the boundary of the stability region. A real characteristic multiplier crosses at the point (1,0) of the complex plane as P approaches the linear segment of the boundary of the stability region, while a complex conjugate pair crosses the unit circle of the complex plane as P approaches the parabolic part of the boundary of the stability region. The characteristic multiplier, μ , of a delay equation is defined as $\mu = e^{\lambda \tau}$, where λ is a root of the characteristic equation. The angular frequency ω of the oscillation at the loss of stability is the imaginary part of the characteristic root with zero real part. Since

$$\lambda = \frac{1}{\tau} \ln \mu = \frac{1}{\tau} \left(\ln |\mu| + i \left(\Theta + 2k\pi \right) \right),$$

where Θ is the angle of μ and k is an integer, then it follows that $\Re \lambda = 0$ if $|\mu| = 1$ and

$$\omega = \Im \lambda = \frac{1}{\tau} \Theta = \frac{1}{\tau} \arctan \frac{\Im \mu}{\Re \mu}$$

as we choose k = 0. Note that $|\mu| = 1$ corresponds to $\Re \nu = 0$, so $\nu = i\kappa$ and $\mu = (i\kappa + 1) / (i\kappa - 1) = (1 - \kappa^2) / (-1 - \kappa^2) + 2\kappa i / (-1 - \kappa^2)$. We have $\Im \mu / \Re \mu = 2\kappa / (1 - \kappa^2)$ and therefore

$$\omega = \frac{1}{\tau} \arctan \frac{2\kappa}{1 - \kappa^2}.$$
(4.12)



Figure 4.2: (a) The critical time delay vs. the length of the pendulum and the spring stiffness (b) The frequency of the oscillation at the loss of stability and the stability domain

Choosing control parameters from the boundary of the stability domain, substituting $\nu = i\kappa$ in the characteristic equation, and solving either its real or imaginary part for κ , the frequency of the oscillation is obtained dividing equation (4.12) by 2π . Since $\arctan 2\kappa / (1 - \kappa^2) < \pi/2$, we have the estimate

$$f < \frac{1}{4\tau} \tag{4.13}$$

for the frequency of oscillation. This means that the frequency of the self-excited oscillation arising after the loss of stability can vary between zero and 25% of the sampling frequency $1/\tau$. The frequency f of the oscillation at the loss of stability is drawn in Figure 4.2(b).

If we have backlash in the system we use (3.2)-(3.3) instead of (2.18). The delayed control force (4.1) leads to the following systems of delay differential

equations

$$\begin{pmatrix} \frac{1}{2} (m+M) m_m r_m & -\frac{1}{4} m m_m l r_m \frac{r_w}{R_w} \\ 0 & \frac{1}{3} m l^2 - \frac{1}{4} \frac{m}{(m+M)} m l^2 \end{pmatrix} \begin{pmatrix} \ddot{\Delta} (t) \\ \ddot{\varphi} (t) \end{pmatrix} \\ + \begin{pmatrix} (m+M) \frac{K}{r_m} & 0 \\ 0 & 0 \end{pmatrix} \begin{pmatrix} \dot{\Delta} (t) \\ \dot{\varphi} (t) \end{pmatrix} \\ + \begin{pmatrix} (m+M) r_m s + \frac{1}{2} m_m r_m \frac{r_w^2}{R_w^2} s & 0 \\ \frac{1}{2} \frac{m}{m+M} l \frac{r_w}{R_w} s & -\frac{1}{2} m g l \end{pmatrix} \begin{pmatrix} \Delta (t) \\ \varphi (t) \end{pmatrix}$$
(4.14)
$$= \begin{pmatrix} (m+M) (P\varphi((j-1)\tau) + D\dot{\varphi}((j-1)\tau)) \\ + (m+M+\frac{1}{2} m_m \frac{r_w}{R_w^2}) r_m s r_0 \operatorname{sgn} \Delta(t) \\ \frac{1}{2} \frac{m}{m+M} l \frac{r_w}{R_w} s r_0 \operatorname{sgn} \Delta(t) \end{pmatrix},$$

for $t \in [j\tau, (j+1)\tau)$ and $j = 1, 2, \ldots$, if $|\Delta| \ge r_0$; and

for j = 1, 2, ..., if $|\Delta| < r_0$. Transform the second order system (4.14) and (4.15) into a system of first order equations. We obtain

$$\begin{split} \dot{\mathbf{y}}\left(t\right) = & \mathbf{A}\mathbf{y}\left(t\right) + \mathbf{b}Q\left(t\right) + \mathbf{f}\\ & Q\left(t\right) = & \mathbf{K}\mathbf{y}\left(\left(j-1\right)\tau\right), \end{split}$$

for $t \in [j\tau, (j+1)\tau)$, and j = 1, 2, ...; and

$$\begin{aligned} \dot{\mathbf{y}}\left(t\right) = \mathbf{A}^{*}\mathbf{y}\left(t\right) + \mathbf{b}Q\left(t\right) \\ Q\left(t\right) = \mathbf{K}\mathbf{y}\left(\left(j-1\right)\tau\right), \end{aligned}$$

for $t \in [j\tau, (j+1)\tau)$ and $j = 1, 2, \ldots$, respectively, where $\mathbf{y}, \mathbf{A}, \mathbf{b}$ are given in (4.4) and \mathbf{K} is given in (4.5), while \mathbf{f} and \mathbf{A}^* assume the form

$$\mathbf{f} = \begin{pmatrix} 0 & \left(\frac{4r_w^2 s r_0}{(m+4M)R_w^2} + \frac{2s r_0}{m_m}\right) \operatorname{sgn} \Delta & 0 & \frac{6r_w s r_0}{(m+4M)lR_w} \operatorname{sgn} \Delta \end{pmatrix}^T, \\ \mathbf{A}^* = \begin{pmatrix} 0 & 1 & 0 & 0\\ 0 & -\frac{2K}{m_m r_m^2} & \frac{3mg r_w}{(m+4M)R_w} & 0\\ 0 & 0 & 0 & 1\\ 0 & 0 & \frac{6(m+M)g}{(m+4M)l} & 0 \end{pmatrix}.$$

The discrete-time model, associated with (4.14) and (4.15), is introduced in the form $\mathbf{r}_{\mathbf{r}_{i}} = \mathbf{A} \cdot \mathbf{r}_{i} + \mathbf{b} \cdot \mathbf{O} + \mathbf{f}_{i}$

$$\mathbf{y}_{j+1} = \mathbf{A}_{\mathbf{d}} \mathbf{y}_j + \mathbf{b}_{\mathbf{d}} Q_j + \mathbf{f}_{\mathbf{d}}$$

$$Q_{j+1} = \mathbf{K} \mathbf{y}_j$$
(4.16)



Figure 4.3: $\dot{\Delta} - \Delta$ phase-plane (a) P = 1 [Nm], D = 0.2 [Nms] (b) P = 10 [Nm], D = 0.5 [Nms]

for j = 1, 2, ..., if $|\Delta| \ge r_0$; and

$$\mathbf{y}_{j+1} = \mathbf{A}_{\mathbf{d}}^* \mathbf{y}_j + \mathbf{b}_{\mathbf{d}} Q_j$$

$$Q_{j+1} = \mathbf{K} \mathbf{y}_j$$
(4.17)

for j = 1, 2, ..., if $|\Delta| < r_0$, respectively, where \mathbf{A}_d and \mathbf{b}_d are given in (4.8), while the constant term \mathbf{f}_d and the matrix \mathbf{A}_d^* is obtained as follows

$$\mathbf{f}_{\mathbf{d}} = \int_{0}^{\tau} e^{\mathbf{A}(\tau-s)} ds \mathbf{f}, \quad \mathbf{A}_{\mathbf{d}}^{*} = e^{\mathbf{A}^{*}\tau}.$$

System (4.16) and (4.17) can be rewritten in the form

$$\begin{aligned} \mathbf{z}_{j+1} = \mathbf{S}\mathbf{z}_j + \mathbf{f}_j \quad |\Delta| \ge r_0 \\ \mathbf{z}_{j+1} = \mathbf{S}^* \mathbf{z}_j \quad |\Delta| < r_0, \end{aligned}$$
(4.18)

for $j = 1, 2, \ldots$, where \mathbf{z}_j and \mathbf{S} is given in (4.10), while

$$\mathbf{S}^* = \begin{pmatrix} \mathbf{A}_{\mathbf{d}}^* & \mathbf{b}_{\mathbf{d}} \\ \mathbf{K} & 0 \end{pmatrix}, \quad \mathbf{f}_j = \begin{pmatrix} \mathbf{f}_{\mathbf{d}} \\ 0 \end{pmatrix}, \quad j = 1, 2, \dots$$

Considering system (4.18), two kinds of stable motions are obtained using MATLAB simulations. The equilibria $(\Delta, \dot{\Delta}, \varphi, \dot{\varphi}) = (\pm r_0, 0, 0, 0)$ are stable left to the dotted line in Figure 4.1(b). A trajectory in the $\dot{\Delta} - \Delta$ phaseplane is drawn in Figure 4.3(a) for P = 1 [Nm], D = 0.2 [Nms]. Since the phase-space is symmetric, the same kind of motion exists around the other equilibrium. A different type of stable motion occurs right to the dotted line in Figure 4.1(b). P = 10 [Nm], D = 0.5 [Nms] in Figure 4.3(b), where this type of motion is shown. Trajectories approach and penetrate a stable set, but since there exists no limit cycle inside the stable set, the motions appear irregular (i.e. quasiperiodic or even chaotic). Peaks can be seen along the trajectories of both motion. They appear at each sampling, when the control force changes.



Figure 4.4: Poincarè maps at the enter to the zone of backlash at $\Delta = r_0$ (a) The 3 dimensional Poincarè map (b), (c), (d) Series of 1 dimensional Poincarè maps

Note that the presented phase-diagrams are results of simulations, thus further studies could provide additional information about the type of motions obtained. In particular, Poincarè maps, Fourier spectra, behavior of nearby trajectories and Lyapunov exponents are investigated numerically with MATLAB in order to learn more about these motions. For the remainder of this section, we used fixed values of the control parameters, i.e., P = 10 [Nm] and D = 0.5 [Nms].

Poincarè maps are shown in Figure 4.4. The three dimensional space $\Delta = r_0 = 1$ [mm] intersects the four dimensional phase-space. For chaotic motions the Poincarè map shows irregularly spread points inside the attractor in the cross section. Figure 4.4(a) shows this three dimensional cross section. Points are spread, but some regularity can be explored. One dimensional Poincarè maps are also drawn, in Figure 4.4(b), 4.4(c) and 4.4(d). Points are situated in an attractor, but the longer we let the simulations run, the denser the filling of the attractor become.

The Fourier spectrum plotted in Figure 4.5(a) is obtained by applying the Fast Fourier Transformation. It shows the highest peak clearly, but many other peaks also appear. It implies that although the motion has a dominating frequency, many other frequencies arise, so the motion is more complicated than periodic.

Nearby trajectories of chaotic motions stretch each other. It means that



Figure 4.5: (a) Fourier spectrum (b) The deviation between nearby trajectories

two trajectories with arbitrarily close initial conditions may move arbitrarily far from each other inside the attractor. Figure 4.5(b) draws the difference between the $\dot{\Delta}$ components of two trajectories in time. The initial difference is 0.001 [m/s]. We can state that trajectories neither approach nor move away from each other, their distance changes between 0 and an upper limit.

The Lyapunov exponents describe the stretching rate of trajectories in a direction. Negative Lyapunov exponent shows that trajectories approach each other in that direction, while positive Lyapunov exponent indicates that trajectories stretch each other in that direction. The Lyapunov exponents are determined by applying the following algorithm [1]. We choose an orthogonal basis $\{\mathbf{w}_1^0, \ldots, \mathbf{w}_5^0\}$ and compute the vectors $\mathbf{v}_1^1, \ldots, \mathbf{v}_5^1$ by multiplying the chosen basis by the matrix \mathbf{S} or \mathbf{S}^* of (4.18) at the initial condition. Use the Gram-Schmidt orthogonalization procedure to orthogonalize these vectors and get $\mathbf{w}_1^*, \ldots, \mathbf{w}_5^*$. To eliminate the problem of extremely large and small numbers, we normalize these vectors to obtain the orthogonal basis of the next step $\{\mathbf{w}_1^1, \ldots, \mathbf{w}_5^1\}$. Repeating this step n times, we get the following estimate for the *i*th Lyapunov exponent

$$\lambda_{i} = \frac{\ln \|\mathbf{w}_{i}^{*n}\| + \dots + \ln \|\mathbf{w}_{i}^{*1}\|}{n}, \quad i = 1, \dots, 5$$

The result of this algorithm shows that three of the Lyapunov exponents are negative, but two of them tend to 0, as n increases. This result predicts that the largest Lyapunov exponent is 0 which means that trajectories neither approach nor move away from each other. This shows good agreement with the discussion in the previous paragraph.

5 Experimental observations

A pendulum-cart structure was constructed [22] and experiments to validate our simulation results were carried out. Measurements were taken with different values of the control parameters P and D and with different time delays τ . In theoretical/simulation studies above we used the assumptions that position



Figure 5.1: The theoretical stability domain and the measured points for $\tau = 5$ [ms]

of the cart was arbitrary, but we had to impose restrictions on the position of the cart in the experimental setting. The proportional and the differential gain of the cart were different from 0, but they were fixed, and those of the pendulum were varied. Therefore, the stability domain in the plane of the control parameters P and D was different from that gained from the system investigated in this paper.

The time delay is $\tau = 5$ [ms], the proportional and the differential gain of the cart are $P_x = 7.7$ [N] and $D_x = 15.4$ [Ns], respectively, in Figure 5.1. The continuous curve surrounds the analitically determined stability domain, while the small circles, squares, crosses and triangle represent measurement points. The circles indicate stable controllers. Since in reality the backlash is always positive and all the disturbances cannot be eliminated, the pendulum oscillates around its upper equilibrium with small amplitude. Figure 5.2(a) shows the time evolution of the angle of the pendulum for P = 100 [Nm], D = 6 [Nms]. The oscillation frequency is about 0.45 [Hz]. The squares in Figure 5.1 indicate controllers at the boundary of the stability domain, where oscillations cause loss of stability. We took measurements for P = 700 [Nm], D = 6 [Nms], the time evolution of the angle of the pendulum is depicted in Figure 5.2(b). The pendulum swings with higher frequency around its upper equilibrium. The oscillation frequency is about 12 [Hz] which is still less than 25 % of the sampling frequency $1/\tau$. It corresponds to the estimate given in (4.13). The crosses in Figure 5.1 represent unstable controllers. Figure 5.2(c) shows the time evolution of the voltage of the motor providing the control force for P = 100 [Nm], D = 2[Nms]. This figure does not show increasing values of the voltage, because it cannot exceed its maximum value. The motor voltage reaches its maximum in a few tenth of a second since the controller is unstable and the pendulum loses its stability. The triangle in Figure 5.1 indicates a controller which works with quite large control force, since both the proportional and the differential gain is large, i.e., P = 700 [Nm], D = 23 [Nms]. Figure 5.2(d) shows that the



Figure 5.2: Time evolutions for different controllers (a) P = 100 [Nm], D = 6 [Nms] (b) P = 700 [Nm], D = 6 [Nms] (c) P = 100 [Nm], D = 2 [Nms] (d) P = 700 [Nm], D = 23 [Nms]

voltage of the motor reaches its maximum value quite frequently which leads to oscillations and loss of stability. The experimental observations do not coincide exactly with the theoretical results, but they show good agreement. If the control force too small, then the upper equilibrium of the pendulum is unstable. There is a domain in the plane of the control parameters where the equilibrium is stable. Increasing the control force causes oscillations and loss of stability. The oscillation frequency arising at the loss of stability is less than 25 % of the sampling frequency.

The theoretical critical time delay is $\tau_{cr} = 88$ [ms]. Stable motions is obtained for $\tau = 20$ [ms], but controllers with delay of 30 [ms] do not stabilize the upper equilibrium of the pendulum. The stability domain is quite small if the time delay is greater than 20 [ms], therefore it is quite difficult to find it experimentally or it does not exist due to some unmodeled effects and disturbances.

The frequency of the motion presented in Figure 5.2(a) and 5.2(b) is about 0.45 [Hz] and 12 [Hz]. The largest peaks occur in the Fourier spectra at these values (see Figure 5.3(a) and 5.3(b)).

Conclusions

Dynamics of digitally controlled inverted pendulum on a cart is examined. It is easy to stabilize the upper equilibrium of the simplest model of the pendulumcart system. A stability domain has been found in the plane of the control parameters. Backlash occuring at the driving wheel introduces small ampli-



Figure 5.3: Fourier spectra (a) P = 100 [Nm], D = 6 [Nms] (b) P = 700 [Nm], D = 6 [Nms]

tude periodic oscillations of the controlled pendulum around its upper equilibrium. The combination of backlash and time delays lead to irregular oscillations. (Our studies so far indicates the appearance of motions which are "marginally" chaotic.) We plan to give a more detailed characterization of these irregularities in future projects.

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